

Piedmont Student Launch Team

Preliminary Design Report

November 15, 2016

Overview

- Vehicle Information
- Flight Statistics
- Motor Choice
- Launch Statistics
- Subsystems
- Payload
- Requirements Compliance Plan
- Q&A

Vehicle Info - Summary

- Length: 98 in
- Diameter: 5.5 in (20.4 in at fins)
- Body material: Fiberglass
- Expected weight: 27.5 lbs

Flight Statistics

- Pre-burnout center of gravity: 54.4 in from nose
- Post-burnout center of gravity: 49.2 in from nose
- Center of pressure: 72.6 in from nose
- Static stability margin: 3.27 calibers

Motor Choice

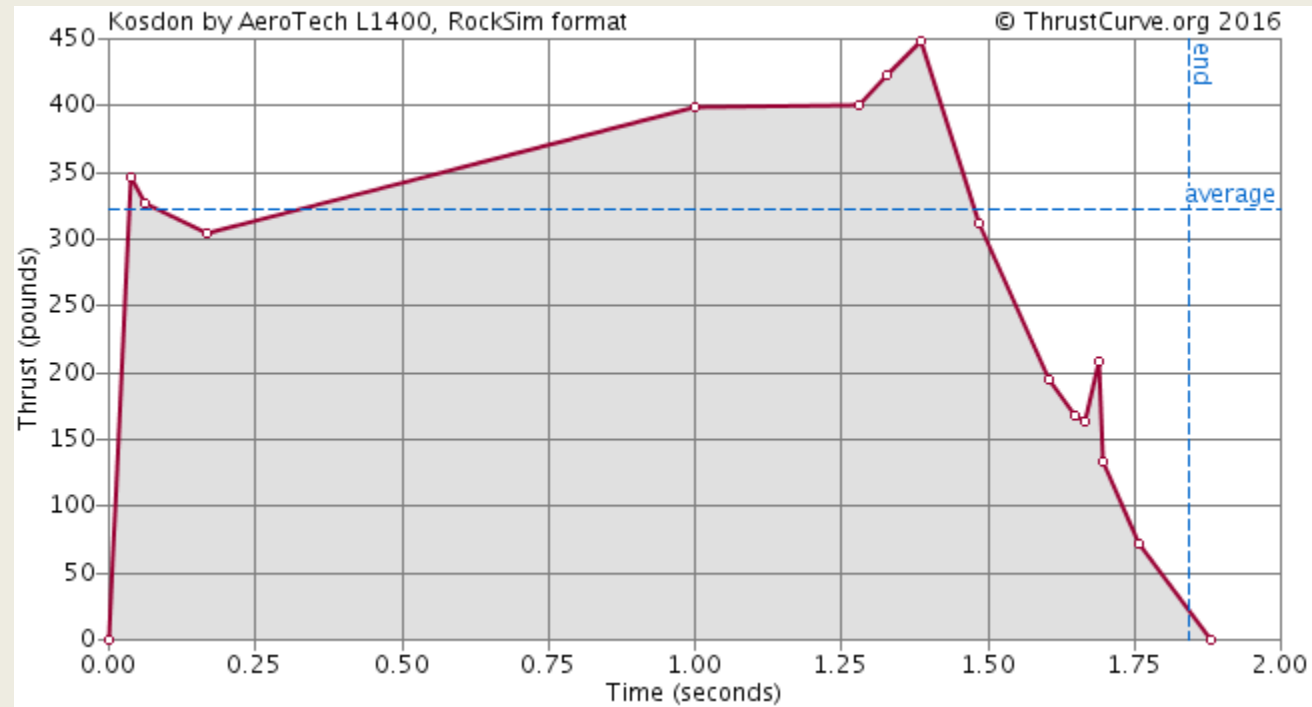
- Criteria:
 - Enough thrust to reach altitude of 5280 ft.
 - Not so much thrust that it requires excessive ballasting.

Motor Choice (cont.)

| Motor | Altitude (ft) |
|---------------------------|---------------|
| Kosdon-by-AeroTech L1400F | 5822.15 |
| Cesaroni L1030RL | 6379.89 |
| Kosdon-by-AeroTech K750W | 5363.68 |

- Kosdon-by-AeroTech L1400F

Motor Choice (cont.)



Launch Statistics

- Thrust to weight ratio: 16.25
- Rail exit velocity: 89.9 ft/s
- Maximum acceleration: 517 ft/s

Vehicle Info - Profile

| Profile | Pros | Cons |
|-------------|---|---|
| Standard | Easy to Design. Easy to Model. Easy to Build. No unusual Aerodynamics | None. |
| Nonstandard | Unique Look. Potentially, more interior space | Unusual aerodynamic effects. Harder to design. Harder to model. Harder to build. |

- Profile refers to the diameter of the rocket and its uniformity.
- Standard



Vehicle Info - Body Sections

| Body Sections | Pros | Cons |
|------------------|--|--|
| Fewer than three | Fewer places for the rocket to separate accidentally | Not enough sections to be able to use a drogue & main parachute w/o more complex recovery. |
| Three | No more parts than needed for recovery system | More time & effort spent manufacturing the body |
| More than three | None | Even more time & effort spent manufacturing the body |

- The body sections refers to the parts of the rocket that are separated by the ejection charges.
- Three sections



Vehicle Info - Bulkheads

| Bulkhead Material | Pros | Cons |
|-------------------|--|---|
| Plywood | Lighter weight. Easier to work with than aluminum. | Not as strong as alternatives. |
| Fiberglass | Easier to work with/lighter than aluminum. | Not as strong as aluminum, more expensive than plywood. |
| Aluminum | Stronger than alternatives. | Most expensive/difficult to work with option. |

- The bulkheads protect the subsystems and act as anchor points.
- Fiberglass

Vehicle Info - Nose Cone

| Nose Cone Shape | Pros | Cons |
|-----------------|--|---|
| Ogive | Commercially available in the same size as the body tubes being used | Somewhat higher drag |
| Parabolic | Somewhat lower drag | Not commercially available at team's body tube size |
| Cone | Lower drag at trans and supersonic speeds | Not commercially available at team's body tube size |
| Elliptical | Somewhat lower drag | Not commercially available at team's body tube size |

- Ogive



Vehicle Info - Altitude Control

| Altitude Control | Pros | Cons |
|------------------|---|--|
| Air Brakes | Precise control of the altitude of the launch Vehicle. Ability to account for variance in launch day conditions. | Need for control electronics. Additional points of mechanical/program failure. Creates turbulent flow/instability. |
| Ballasting | Easy to Manufacture. Few extra points of failure. No exterior parts. | Less precise altitude control. Cannot account for launch day conditions. |

- Necessary system for the altitude challenge. Allows for accurate arrival at specific altitudes.

- Ballasting



Vehicle Info - Fins

| Fin shape | Pros | Cons | Number of fins | Pros | Cons |
|-------------|---|---|----------------|--|---|
| Trapezoidal | Easy design. Easy manufacture. Easier to attach | Higher drag than elliptical. | Three | Less weight. Less drag. Less time/effort/resources. Less likely to be unstable. | None. |
| Elliptical | Less drag than trapezoidal. | Difficult to manufacture. Difficult to design. | Four | None | More weight. More drag. More time/effort/resources. |
| Tube | Unique Look. | Little information to base decisions on. | | | |

- Trapezoidal

- Three fins



Recovery System - Components

- Parachutes
 - Drogue Chute
 - Main Chute
- Harnesses
 - Kevlar
- Redundancy
 - Dual Ejection Charges/Altimeters

Recovery System - Strategy

- Deployment of rocket
 - Altimeters observe stalling height reached for the rocket
 - Drogue chute deployed to reach intended altitude
- Apogee reached
 - Drogue chute partially slows descent
- Recovery altitude reached
 - Altimeters observe recovery height has been reached on descent
 - Main parachute deploys, allowing easier recovery of rocket

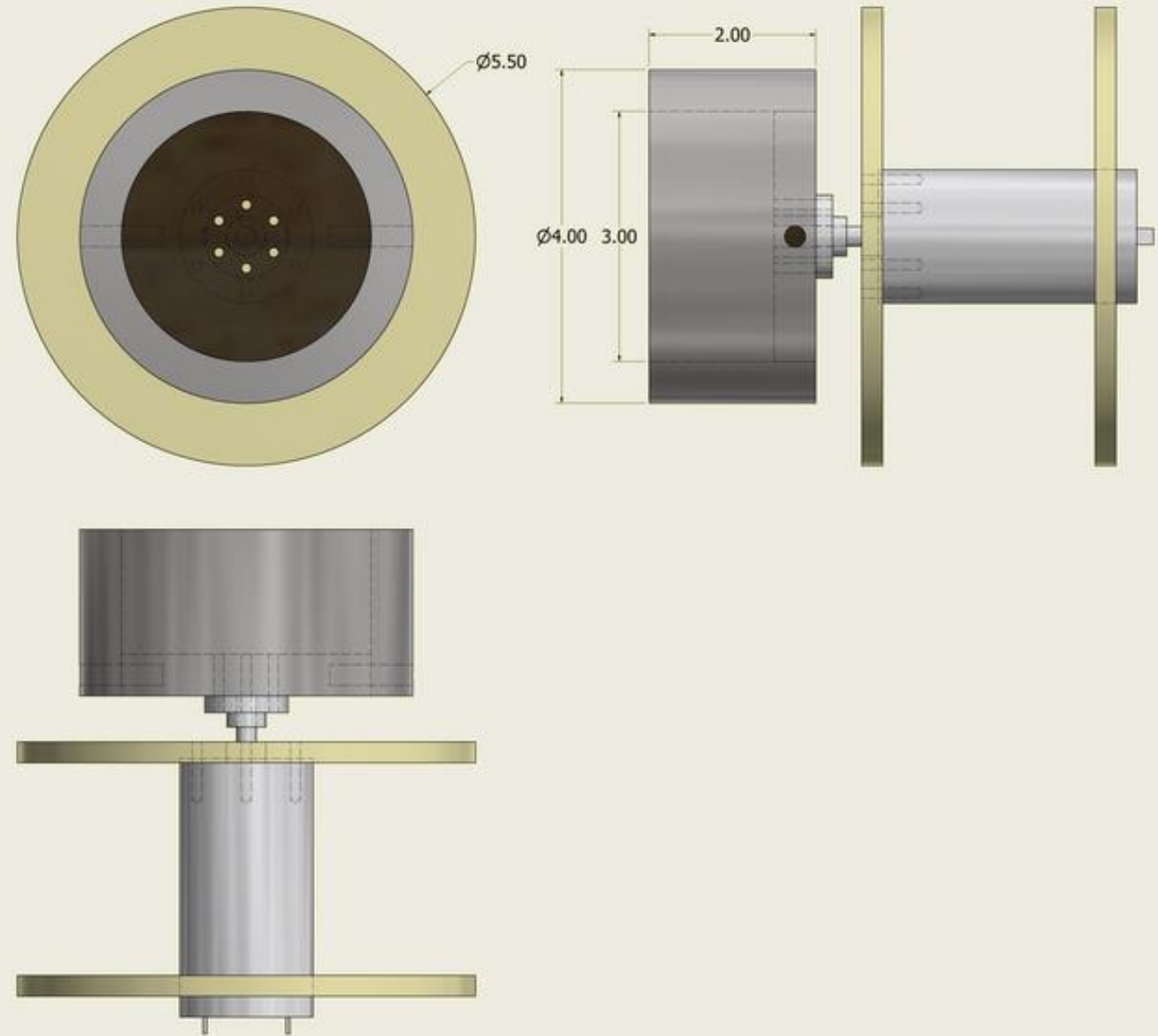
Payload Summary

- Roll and counter-roll
- Reaction wheel
 - Conservation of Angular Momentum
 - Precise control through ascent
- Team criteria
 - Identify ground targets
 - Aim camera
 - Transmit video

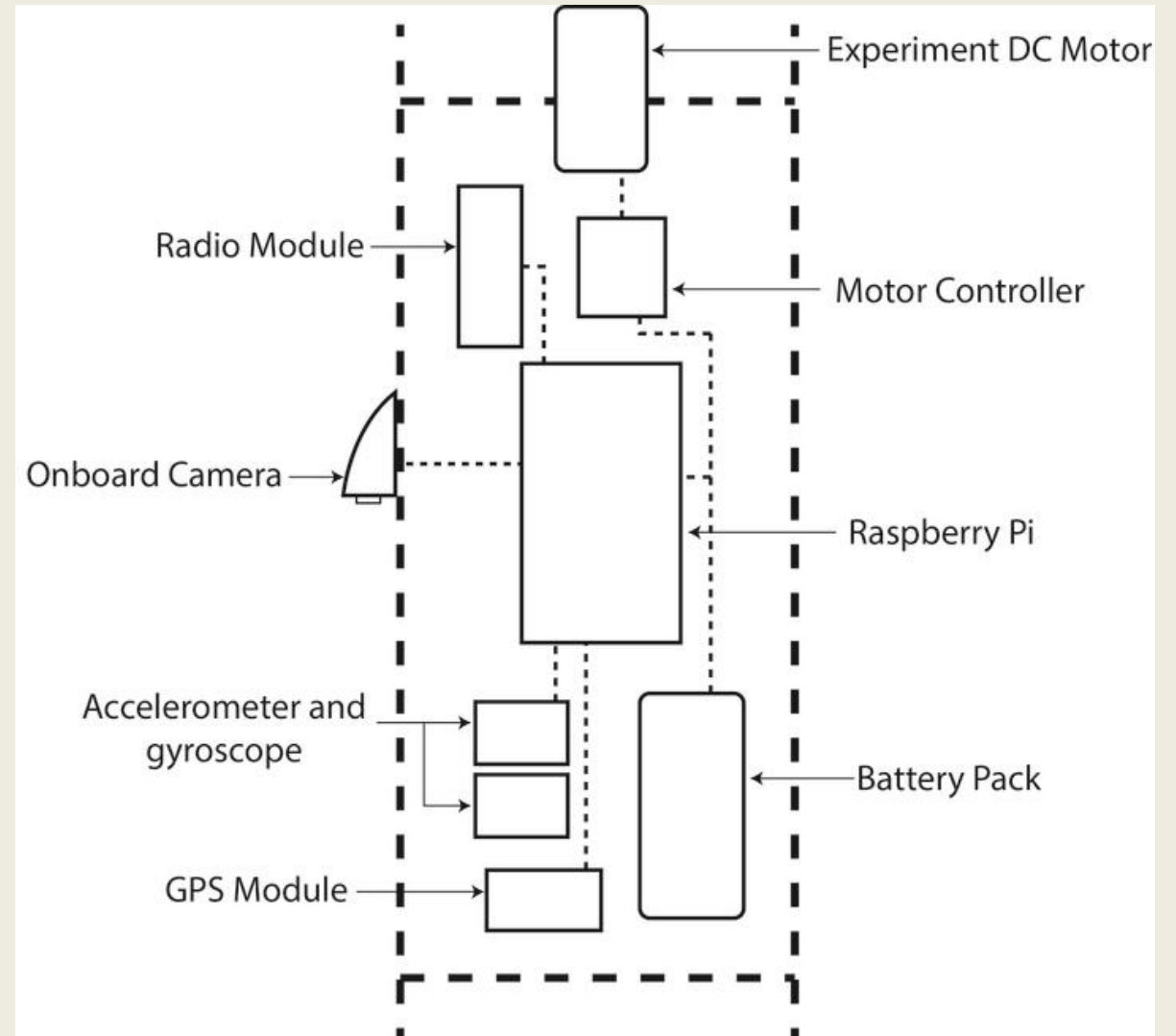
Payload Events

- Detect motor burnout
- Measure rotation
- Rotate twice
- Rotate to targets
- Return to burnout rotation
- Transmit data and video

Reaction Wheel

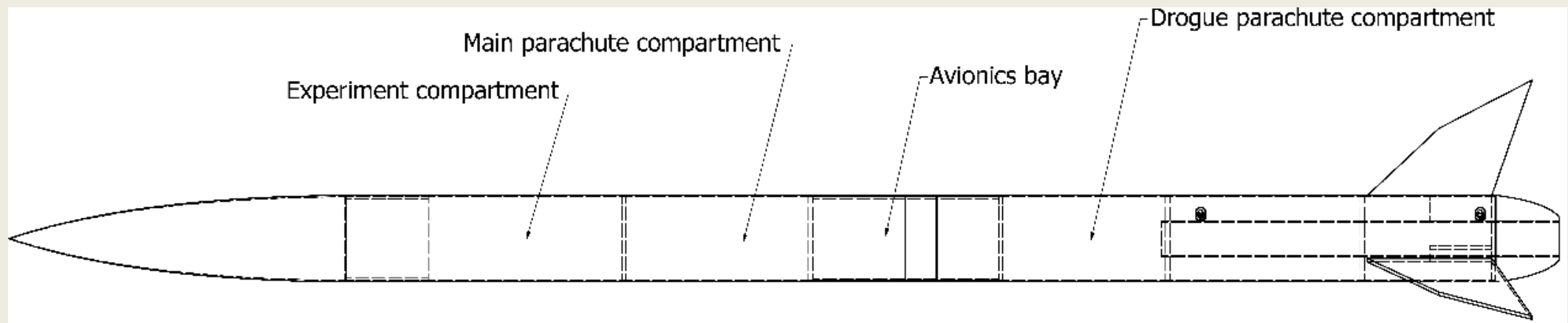


Payload Electronics



Payload Placement

- Toward nose
- Moves center of gravity forward
- Separated from recovery system



Data Transmission

- 900 MHz radio
- Data
 - GPS position
 - Velocity
 - Acceleration
 - Rotation
- Video

Requirements Compliance Plan – Launch Vehicle

- Altitude challenge
 - Two altimeters
 - Ballast
 - Testing
- Propulsion system
 - Single, ammonium perchlorate composite solid motor
- Subscale rocket
 - Design is happening
 - Construction will begin soon
 - Launch will happen around December 10th

Requirements Compliance Plan – Launch Vehicle (cont.)

- Full scale test flight
 - Final design is happening
 - Construction of the full scale rocket will begin after CDR
 - Launch will happen in mid to late February

Requirements Compliance Plan – Recovery System

- Parachutes
 - Drogue
 - Main
- Test fire
 - Will happen in early to mid February
- Electronics
 - Same altimeters as altitude challenge
 - Avionics bay will be shielded
- Ejection charges
 - Each altimeter will fire a separate ejection charge for the drogue and main parachutes

Requirements Compliance Plan – Recovery System (cont.)

- Tracking system
 - Will have a GPS
 - Will transmit the location of the rocket along with other flight data

Requirements Compliance Plan – Experiment

- Roll induction
 - Will use a reaction wheel
- Proof of success
 - Will use cameras, accelerometers, and gyroscopes
- Team requirements
 - Will be programmed to turn the rocket to aim a camera at one of the ground targets
 - Will transmit live video from a camera

Requirements Compliance Plan – Safety

- Safety officer
 - Nicolas
- Mentor
 - Review all designs to ensure they are safe
 - Instruct team members in safe and effective rocket design
- Safety plan
 - Come up with check lists for launches
 - Oversee all construction
 - Review all construction documents to ensure safety procedures and hazard warnings are in place

Requirements Compliance Plan – General

- Project plan
 - A project planning document is available to the entire team and is updated by the project manager
 - The project plan includes timeline, requirements, budget, and other information
- Educational engagement
 - PSLT is making a specific effort to do educational engagement
 - There are already several activities planned
 - PSLT has decide to engage a minimum of 200 women and girls in educational activities

Requirements Compliance Plan – General (cont.)

- Website
 - The team has a webmaster
 - There is a page on the website for documents

- Teleconference set up
 - Available for all future meetings
 - Provided by our sponsors

Requirements Compliance Plan – Other

- The project manager will ensure that all of the requirements of the project are met
- Sub-team leaders will ensure the requirements for their parts of the project are met

Questions?